

**SYLLABUS****M E-452: Dynamics of Space Flight****Fall 2016**Text: *Orbital Mechanics for Engineering Students*, H. D. Curtis (Elsevier, 2010) 2<sup>nd</sup> Edition

References: Additional materials will be e-mailed through the Highlander Pipeline during the semester

<b>Week</b>	<b>Topic</b>	<b>Reading</b>	<b>Problems</b>
<b>1</b>	Introduction and Review: Kinematics, Newton's Laws, Time derivative of moving vectors, relative motion	Chapter 1	1.1 – 1.10 1.12, 1.13, 1.15
<b>2</b>	Two Body Problem: Motion equations in an Inertial Frame; Relative Motion	2.1 - 2.3	
<b>3</b>	Review of Conic sections; Angular Momentum Orbital Energy	Electronic notes; 2.4, 2.5	
<b>4</b>	Derivation of Equations for Orbital Trajectories: Circular, Parabolic, Elliptic  Hyperbolic Trajectories	2.6 – 2.9	2.1-2.3, 2.6, 2.7, 2.8, 2.11, 2.13, 2.15-2.18, 2.21, 2.22, 2.24, 2.30, 2.32 2.37, 2.38
<b>5</b>	<b>Exam 1</b>	2.10	
<b>6</b>	Perifocal Frame  Restricted Three-Body Problem	2.10  2.11	2.26, 2.32, 2.35, 2.36 2.44, 2.45
<b>7</b>	Orbital Position as a Time Function: Time since Periapsis- Circular Elliptical Trajectories	3.1, 3.2 3.5	3.1, 3.4 - 3.10 3.14
<b>8</b>	Hyperbolic Trajectories Universal Variables	3.6 3.7	3.16 3.19
<b>9</b>	<b>Exam 2</b>		
<b>10</b>	Orbits in 3 Dimensions: State vector & Geocentric Frame Orbital Elements and State Vector	4.1 – 4.3 4.4	4.1, 4.2 4.4, 4.5
<b>11</b>	Coordinate Transformations Transformation: Geocentric Equatorial to Perifocal Frames	4.5 4.6	4.8 – 4.11 4.14 – 4.16, 4.19,
<b>12</b>	Effects of Earth's Oblateness	4.7	4.25, 4.26
<b>13</b>	<b>Exam 3</b>		
<b>14</b>	Preliminary Orbit Determination: Gibbs' Method Lambert's Problem  Review of Course Concepts	5.1, 5.2 5.3	5.1, 5.2 5.4, 5.5
<b>15</b>	<b>Final Exam</b>		

\* Homework Problems will be assigned in class.